

Natural frequencies of double section beam, with mass at the top

$M_{trolley} := 3300 \text{ kg}$ suspended mass

$l := 0.01 \text{ m}$ position of suspended mass from the top

$A_1 := 192 \text{ cm}^2$ bottom column cross section area

$I_1 := 65810 \text{ cm}^4$ bottom column cross section inertia

$l_1 := 7.4 \text{ m}$ bottom column lenght

$A_2 := 192 \text{ cm}^2$ top column cross section area

$I_2 := 65810 \text{ cm}^4$ top column cross section inertia

$l_2 := 7.4 \text{ m}$ top column lenght

$E := 206000 \frac{\text{N}}{\text{mm}^2}$ Hooke modulus

$\rho := 7850 \frac{\text{kg}}{\text{m}^3}$ density



$\alpha := \frac{l_2 - l}{l_2} = 0.998648648648649$

ELEMENTS STIFFNESS MATRIX

$$K_{11} := \frac{\frac{E}{2} \cdot \frac{I_1}{m^4}}{\left(\frac{l_1}{2}\right)^3} \cdot \begin{bmatrix} 12 & 6 \cdot \frac{l_1}{2} & -12 & 6 \cdot \frac{l_1}{2} \\ 6 \cdot \frac{l_1}{2} & 4 \cdot \left(\frac{l_1}{2}\right)^2 & -6 \cdot \frac{l_1}{2} & 2 \cdot \left(\frac{l_1}{2}\right)^2 \\ -12 & -6 \cdot \frac{l_1}{2} & 12 & -6 \cdot \frac{l_1}{2} \\ 6 \cdot \frac{l_1}{2} & 2 \cdot \left(\frac{l_1}{2}\right)^2 & -6 \cdot \frac{l_1}{2} & 4 \cdot \left(\frac{l_1}{2}\right)^2 \end{bmatrix}$$

$$K_{12} := \frac{\frac{E}{2} \cdot \frac{I_1}{m^4}}{\left(\frac{l_1}{2}\right)^3} \cdot \begin{bmatrix} 12 & 6 \cdot \frac{l_1}{2} & -12 & 6 \cdot \frac{l_1}{2} \\ 6 \cdot \frac{l_1}{2} & 4 \cdot \left(\frac{l_1}{2}\right)^2 & -6 \cdot \frac{l_1}{2} & 2 \cdot \left(\frac{l_1}{2}\right)^2 \\ -12 & -6 \cdot \frac{l_1}{2} & 12 & -6 \cdot \frac{l_1}{2} \\ 6 \cdot \frac{l_1}{2} & 2 \cdot \left(\frac{l_1}{2}\right)^2 & -6 \cdot \frac{l_1}{2} & 4 \cdot \left(\frac{l_1}{2}\right)^2 \end{bmatrix}$$

$$\left[\begin{array}{cc} 6 \cdot \frac{\frac{l}{2}}{m} & 2 \cdot \left(\frac{\frac{l}{2}}{m} \right) \\ -6 \cdot \frac{\frac{l}{2}}{m} & 4 \cdot \left(\frac{\frac{l}{2}}{m} \right) \end{array} \right]$$

$$K_{21} := \frac{\frac{E}{m} \cdot \frac{I_2}{2} \cdot \frac{1}{m^4}}{\left(\frac{\alpha \cdot l_2}{m} \right)^3} \cdot \left[\begin{array}{cccc} 12 & 6 \cdot \frac{\alpha \cdot l_2}{m} & -12 & 6 \cdot \frac{\alpha \cdot l_2}{m} \\ 6 \cdot \frac{\alpha \cdot l_2}{m} & 4 \cdot \left(\frac{\alpha \cdot l_2}{m} \right)^2 & -6 \cdot \frac{\alpha \cdot l_2}{m} & 2 \cdot \left(\frac{\alpha \cdot l_2}{m} \right)^2 \\ -12 & -6 \cdot \frac{\alpha \cdot l_2}{m} & 12 & -6 \cdot \frac{\alpha \cdot l_2}{m} \\ 6 \cdot \frac{\alpha \cdot l_2}{m} & 2 \cdot \left(\frac{\alpha \cdot l_2}{m} \right)^2 & -6 \cdot \frac{\alpha \cdot l_2}{m} & 4 \cdot \left(\frac{\alpha \cdot l_2}{m} \right)^2 \end{array} \right]$$

$$K_{22} := \frac{\frac{E}{m} \cdot \frac{I_2}{2} \cdot \frac{1}{m^4}}{\left(\frac{(1-\alpha) \cdot l_2}{m} \right)^3} \cdot \left[\begin{array}{cccc} 12 & 6 \cdot \frac{(1-\alpha) \cdot l_2}{m} & -12 & 6 \cdot \frac{(1-\alpha) \cdot l_2}{m} \\ 6 \cdot \frac{(1-\alpha) \cdot l_2}{m} & 4 \cdot \left(\frac{(1-\alpha) \cdot l_2}{m} \right)^2 & -6 \cdot \frac{(1-\alpha) \cdot l_2}{m} & 2 \cdot \left(\frac{(1-\alpha) \cdot l_2}{m} \right)^2 \\ -12 & -6 \cdot \frac{(1-\alpha) \cdot l_2}{m} & 12 & -6 \cdot \frac{(1-\alpha) \cdot l_2}{m} \\ 6 \cdot \frac{(1-\alpha) \cdot l_2}{m} & 2 \cdot \left(\frac{(1-\alpha) \cdot l_2}{m} \right)^2 & -6 \cdot \frac{(1-\alpha) \cdot l_2}{m} & 4 \cdot \left(\frac{(1-\alpha) \cdot l_2}{m} \right)^2 \end{array} \right]$$

SYSTEM STIFFNESS MATRIX

```
merge (base, data, row, col) :=  $\left\{ \begin{array}{l} out := base \\ \text{for } r \in [1..rows(data)] \\ \quad \text{for } c \in [1..cols(data)] \\ \quad \quad out_{row+r-1, col+c-1} := data_{r, c} \end{array} \right.$ 
function to
assembly matrix
out
```

$$K := matrix(10, 10) = \left[\begin{array}{cccccccccc} 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \end{array} \right]$$

$$\begin{bmatrix} 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \end{bmatrix}$$

$$K_{system} := merge \left(K, K_{11} [1..4][1..4], 1, 1 \right) + merge \left(K, K_{12} [1..4][1..4], 3, 3 \right) + merge \left(K, K_{21} [1..4][1..4], 3, 3 \right) + merge \left(K, K_{22} [1..4][1..4], 3, 3 \right)$$

CONSTARINED SYSTEM STIFFNESS MATRIX

$$m := rows \left(K_{system} \right) = 10$$

$$K_{system_cons} := submatrix \left(K_{system}, 3, m, 3, m \right)$$

ELEMENTS MASS MATRIX

$$M_{11} := \frac{\frac{\rho}{\frac{kg}{m^3}} \cdot \frac{A_1}{m^2} \cdot \frac{l_1}{2}}{420} \cdot \begin{bmatrix} 156 & 22 \cdot \frac{l_1}{2} & 54 & -13 \cdot \frac{l_1}{2} \\ 22 \cdot \frac{l_1}{2} & 4 \cdot \left(\frac{l_1}{2} \right)^2 & 13 \cdot \frac{l_1}{2} & -3 \cdot \left(\frac{l_1}{2} \right)^2 \\ 54 & 13 \cdot \frac{l_1}{2} & 156 & -22 \cdot \frac{l_1}{2} \\ -13 \cdot \frac{l_1}{2} & -3 \cdot \left(\frac{l_1}{2} \right)^2 & -22 \cdot \frac{l_1}{2} & 4 \cdot \left(\frac{l_1}{2} \right)^2 \end{bmatrix}$$

$$M_{12} := \frac{\frac{\rho}{\frac{kg}{m^3}} \cdot \frac{A_1}{m^2} \cdot \frac{l_1}{2}}{420} \cdot \begin{bmatrix} 156 & 22 \cdot \frac{l_1}{2} & 54 & -13 \cdot \frac{l_1}{2} \\ 22 \cdot \frac{l_1}{2} & 4 \cdot \left(\frac{l_1}{2} \right)^2 & 13 \cdot \frac{l_1}{2} & -3 \cdot \left(\frac{l_1}{2} \right)^2 \\ 54 & 13 \cdot \frac{l_1}{2} & 156 & -22 \cdot \frac{l_1}{2} \\ -13 \cdot \frac{l_1}{2} & -3 \cdot \left(\frac{l_1}{2} \right)^2 & -22 \cdot \frac{l_1}{2} & 4 \cdot \left(\frac{l_1}{2} \right)^2 \end{bmatrix}$$

$$\frac{\rho}{\frac{kg}{m^3}} \cdot \frac{A_2}{m^2} \cdot \frac{\alpha \cdot l_2}{m} \cdot \begin{bmatrix} 156 & 22 \cdot \frac{\alpha \cdot l_2}{m} & 54 & -13 \cdot \frac{\alpha \cdot l_2}{m} \\ 22 \cdot \frac{\alpha \cdot l_2}{m} & 4 \cdot \left(\alpha \cdot l_2 \right)^2 & 13 \cdot \frac{\alpha \cdot l_2}{m} & -3 \cdot \left(\alpha \cdot l_2 \right)^2 \\ 54 & 13 \cdot \frac{\alpha \cdot l_2}{m} & 156 & -22 \cdot \frac{\alpha \cdot l_2}{m} \\ -13 \cdot \frac{\alpha \cdot l_2}{m} & -3 \cdot \left(\alpha \cdot l_2 \right)^2 & -22 \cdot \frac{\alpha \cdot l_2}{m} & 4 \cdot \left(\alpha \cdot l_2 \right)^2 \end{bmatrix}$$

$$M_{21} := \frac{m^3}{420} \cdot \begin{bmatrix} m^4 \cdot \left(\frac{m}{m} \right) & m^{-3} \cdot \left(\frac{m}{m} \right) \\ 54 & 13 \cdot \frac{\alpha \cdot l_2}{m} & 156 & -22 \cdot \frac{\alpha \cdot l_2}{m} \\ -13 \cdot \frac{\alpha \cdot l_2}{m} & -3 \cdot \left(\frac{\alpha \cdot l_2}{m} \right)^2 & -22 \cdot \frac{\alpha \cdot l_2}{m} & 4 \cdot \left(\frac{\alpha \cdot l_2}{m} \right) \end{bmatrix}$$

$$M_{22} := \frac{\frac{\rho}{\frac{kg}{3}} \cdot \frac{A_2}{m^2} \cdot \frac{(1-\alpha) \cdot l_2}{m}}{420} \cdot \begin{bmatrix} 156 & 22 \cdot \frac{(1-\alpha) \cdot l_2}{m} & 54 & -13 \cdot \frac{(1-\alpha) \cdot l_2}{m} \\ 22 \cdot \frac{(1-\alpha) \cdot l_2}{m} & 4 \cdot \left(\frac{(1-\alpha) \cdot l_2}{m} \right)^2 & 13 \cdot \frac{(1-\alpha) \cdot l_2}{m} & -3 \cdot \left(\frac{(1-\alpha) \cdot l_2}{m} \right)^2 \\ 54 & 13 \cdot \frac{(1-\alpha) \cdot l_2}{m} & 156 & -22 \cdot \frac{(1-\alpha) \cdot l_2}{m} \\ -13 \cdot \frac{(1-\alpha) \cdot l_2}{m} & -3 \cdot \left(\frac{(1-\alpha) \cdot l_2}{m} \right)^2 & -22 \cdot \frac{(1-\alpha) \cdot l_2}{m} & 4 \cdot \left(\frac{(1-\alpha) \cdot l_2}{m} \right) \end{bmatrix}$$

SYSTEM MASS MATRIX

```
merge (base, data, row, col) := | out := base
                                | for r ∈ [1..rows (data)]
                                |   for c ∈ [1..cols (data)]
                                |     out row + r - 1 col + c - 1 := data r c
                                | out
```

function to
assembly matrix

$$M := \text{matrix}(10, 10) = \begin{bmatrix} 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \end{bmatrix}$$

$$M_{77} := \frac{M_{trolley}}{kg}$$

$$M_{system} := merge \left(M, M_{11} [1..4][1..4], 1, 1 \right) + merge \left(M, M_{12} [1..4][1..4], 3, 3 \right) + merge \left(M, M_{21} [1..4][1..4], 3, 3 \right) + merge \left(M, M_{22} [1..4][1..4], 3, 3 \right)$$

CONSTARINED SYSTEM MASS MATRIX

$$\alpha := \text{rows} (M \quad) = 10$$

$$M_{system_cons} := \text{submatrix}(M_{system}, 3, q, 3, q)$$

DYNAMICAL MATRIX

$$A := \text{invert}(K_{system_cons}) \cdot M_{system_cons}$$

EIGENVALUE EVALUATION

$$MEigen(A) := \left[\begin{array}{l} n := \text{rows}(A) \quad C := A \quad Id := \text{identity}(n) \quad v := \text{matrix}(n, 1) \quad s := v \quad v_1 := \text{tr}(A) \\ E := \text{matrix}(n, n) \quad R := \text{matrix}(n-1, n-1) \quad ro := [1..n] \\ \text{for } k := 2, k \leq n, k := k+1 \\ \quad \left[C := A \cdot (C - v_{k-1} \cdot Id) \quad v_k := \frac{\text{tr}(C)}{k} \right] \\ \lambda := \text{polyroots}(\text{stack}(-\text{reverse}(v), 1)) \\ \text{for } k \in ro \\ \quad \left[\begin{array}{l} kOut(X) := \left[\begin{array}{l} j := 0 \quad W := \text{matrix}(n-1, 1) \\ \text{for } r \in ro \\ \quad \text{if } r \neq k \\ \quad \quad W_r := X_j := j+1 \\ \quad \text{else} \\ \quad \quad W_r := 1 \\ \quad \quad \text{continue} \\ \quad W \end{array} \right] \\ kIn(X) := \left[\begin{array}{l} j := 0 \quad W := \text{matrix}(n, 1) \\ \text{for } r \in ro \\ \quad \text{if } r \neq k \\ \quad \quad W_r := X_j := j+1 \\ \quad \text{else} \\ \quad \quad W_r := 1 \\ \quad \quad W \\ \quad \quad \text{norme}(W) \end{array} \right] \\ v_{ro} := \text{str2num}(\text{concat}("_MEIGENV\#_", \text{num2str}(ro))) \\ \left[u := \text{col}(Id, k) \quad v_k := 1 \quad eq := kOut(A \cdot v - \lambda_k \cdot v) \quad vo := kOut(v) \right] \\ \left[r := [1..(n-1)] \quad c := r \right] \\ R_{rc} := \text{str2num}(\text{concat}("diff(", \text{num2str}(eq_r), ",", \text{num2str}(vo_c), ")")) \\ \left[\begin{array}{l} \varepsilon := \text{try} \\ \quad kIn(-R^{-1} \cdot kOut(A \cdot u - \lambda_k \cdot u)) \left[\begin{array}{l} E_{ro\ k} := \varepsilon_{ro} \\ S_k := \text{norme}(A \cdot \varepsilon - \lambda_k \cdot \varepsilon) \end{array} \right] \\ \text{on error} \\ \quad \text{matrix}(n, 1) \end{array} \right] \end{array} \right] \\ \left[\lambda \quad E \right] \end{array} \right]$$

$$[\lambda \quad M] := MEigen(\text{eval}(A)) = \left[\begin{array}{l} \left[\begin{array}{c} -9.6786 \cdot 10^{-9} \\ 9.6786 \cdot 10^{-9} \\ 1.3878 \cdot 10^{-17} + 9.0295 \cdot 10^{-9} \cdot i \\ 1.3878 \cdot 10^{-17} - 9.0295 \cdot 10^{-9} \cdot i \\ -4.4286 \cdot 10^{-9} \\ 1.6411 \cdot 10^{-5} \\ 0.0002 \\ 0.0567 \end{array} \right] \left[\begin{array}{cccc} 0.4059 & 0.0663 & -0.1511 + 0.0019 \cdot i & 0.1422 + 0.0041 \cdot i \\ 0.4426 & 0.9188 & -0.8666 - 0.0076 \cdot i & 0.6626 + 0.0018 \cdot i \\ -0.4743 - 0.2625 & 0.334 & & -0.3536 - 0.0082 \cdot i \\ 0.2251 & 0.1321 & -0.2004 + 0.001 \cdot i & 0.1146 \\ -0.0247 - 0.0105 & 0.0118 + 7.3823 \cdot 10^{-5} \cdot i & -0.0254 - 0.001 \cdot i & \\ -0.4256 & -0.18 & 0.1924 + 0.0015 \cdot i & -0.4474 - 0.0173 \cdot i \\ -0.0289 - 0.0123 & 0.0137 + 8.9184 \cdot 10^{-5} \cdot i & -0.0299 - 0.0011 \cdot i & \\ -0.4256 & -0.18 & 0.1924 + 0.0015 \cdot i & -0.4474 - 0.0173 \cdot i \end{array} \right] \end{array} \right]$$

norme(A.M - M.diag(lambda)) = 0

or better

[lambda M] := Re (MEigen (eval (A))) = [[-9.6786.10^-9 9.6786.10^-9 1.3878.10^-17 1.3878.10^-17 -4.4286.10^-9 1.6411.10^-5 0.0002 0.0567] [0.4059 0.0663 -0.1511 0.1422 -3.9864.10^-6 0.7331 0.1166 0.059 0.4426 0.9188 -0.8666 0.6626 -1.6639.10^-5 0.1287 0.0476 0.030 -0.4743 -0.2625 0.334 -0.3536 5.7359.10^-5 0.1255 0.2535 0.215 0.2251 0.1321 -0.2004 0.1146 3.855.10^-5 -0.3792 0.0164 0.052 -0.0247 -0.0105 0.0118 -0.0254 0.0001 0.0404 -0.0196 0.684 -0.4256 -0.18 0.1924 -0.4474 0.0003 0.4414 -0.0682 0.068 -0.0289 -0.0123 0.0137 -0.0299 -1 0.0434 0.9539 0.684 -0.4256 -0.18 0.1924 -0.4474 0.0004 0.2969 -0.0682 0.068]]

norme(A.M - M.diag(lambda)) = 0

lambda := dn_LinAlgEigenvalues(A) = [0.0567 0.0002 1.5588.10^-5 2.1646.10^-6 -1.5921.10^-6 1.9853.10^-7 1.1.10^-14 3.2702.10^-16]

M := dn_LinAlgEigenvectors(A) = [[0.0594 -0.3876 0.7642 0.4958 0.019 0.0813 -1.7029.10^-7 -1.7636.10^-6 0.0306 -0.1583 0.1268 -0.3112 0.0173 0.908 -6.5593.10^-7 -6.7948.10^-6 0.2154 -0.8428 0.0845 -0.4867 -0.1408 -0.2773 1.3133.10^-7 1.4177.10^-6 0.0522 -0.0544 -0.4042 0.3519 -0.2257 0.1454 -9.5712.10^-7 -9.848.10^-6 0.6841 0.0649 0.0354 -0.0257 -0.0339 -0.0111 3.4654.10^-7 7.2782.10^-6 0.0689 0.2267 0.3366 -0.3841 -0.6804 -0.188 -9.1124.10^-7 -7.1421.10^-6 0.6848 0.0672 0.0388 -0.0296 -0.0407 -0.013 -0.0051 -0.0671 0.0689 0.2267 0.3366 -0.3841 -0.6804 -0.188 -1 0.9977]]

norme(A.M - M.diag(lambda)) = 7.10^-16

NATURAL FREQUENCIES

n := rows(A)

for j in [1..n] f_j := sqrt(1 / (dn_LinAlgEigenvalues(A)_j * 2 * pi))

[0.6686 11.1367 40.3105 108.1762]

$$f = \begin{bmatrix} -126.1343 \cdot i \\ 357.1928 \\ 1.5175 \cdot 10^6 \\ 8.8011 \cdot 10^6 \end{bmatrix}$$

$$f_1 := f_1 \text{ Hz}$$

$$f_2 := f_2 \text{ Hz}$$

$$f_3 := f_3 \text{ Hz}$$

$$f_4 := f_4 \text{ Hz}$$

CHECKING VALUE: EXACT VALUE OF NATURAL FREQUENCIES (UNIFORM CROSS SECTION BEAM and WITHOUT SUSPENDED MASS OF THE TROLLEY)

$$f_{1exact} := \frac{(1.875)^2}{2 \cdot \pi} \cdot \sqrt{\frac{E \cdot I_1}{\rho \cdot A_1 \cdot (l_1 + l_2)^4}} = 2.4227 \text{ Hz} \qquad \frac{f_1 - f_{1exact}}{f_1} = -262.3251 \%$$

$$f_{2exact} := \frac{(4.694)^2}{2 \cdot \pi} \cdot \sqrt{\frac{E \cdot I_1}{\rho \cdot A_1 \cdot (l_1 + l_2)^4}} = 15.1837 \text{ Hz} \qquad \frac{f_2 - f_{2exact}}{f_2} = -36.3388 \%$$

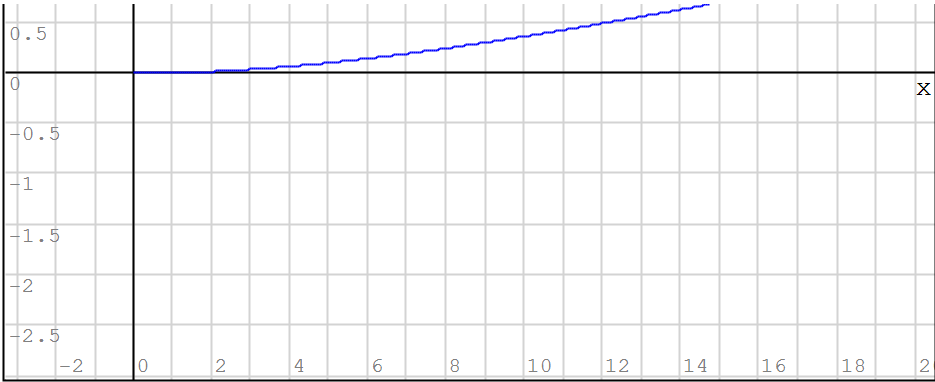
$$f_{3exact} := \frac{(7.855)^2}{2 \cdot \pi} \cdot \sqrt{\frac{E \cdot I_1}{\rho \cdot A_1 \cdot (l_1 + l_2)^4}} = 42.519 \text{ Hz} \qquad \frac{f_3 - f_{3exact}}{f_3} = -5.4788 \%$$

MODE SHAPES

$$w_{11_1}(x) := \left(3 \cdot \left(\frac{x}{\frac{l_1}{2}} \right)^2 - 2 \cdot \left(\frac{x}{\frac{l_1}{2}} \right)^3 \right) \cdot \text{dn_LinAlgEigenvectors}(A)_{11} + \left(- \left(\frac{x}{\frac{l_1}{2}} \right)^2 + \left(\frac{x}{\frac{l_1}{2}} \right)^3 \right) \cdot \frac{\frac{l_1}{2}}{m} \cdot \text{dn_LinAlgEig}$$

$$w_{element1_1}(x) := w_{11_1}(x)$$

$$w_{12_1}(x) := \left(1 - 3 \cdot \left(\frac{x}{\frac{l_1}{2}} \right)^2 + 2 \cdot \left(\frac{x}{\frac{l_1}{2}} \right)^3 \right) \cdot \text{dn_LinAlgEigenvectors}(A)_{11} + \left(\frac{x}{\frac{l_1}{2}} - 2 \cdot \left(\frac{x}{\frac{l_1}{2}} \right)^2 + \left(\frac{x}{\frac{l_1}{2}} \right)^3 \right) \cdot \frac{\frac{l_1}{2}}{m} \cdot \text{dn_}$$



$w_{first}(x)$

$$w_{11_2}(x) := \left(3 \cdot \left(\frac{x}{\frac{l_1}{2}} \right)^2 - 2 \cdot \left(\frac{x}{\frac{l_1}{2}} \right)^3 \right) \cdot \text{dn_LinAlgEigenvectors}(A)_{1\ 2} + \left(- \left(\frac{x}{\frac{l_1}{2}} \right)^2 + \left(\frac{x}{\frac{l_1}{2}} \right)^3 \right) \cdot \frac{\frac{l_1}{2}}{m} \cdot \text{dn_LinAlgEigenvectors}(A)_{2\ 2}$$

$$w_{element1_2}(x) := w_{11_2}(x)$$

$$w_{12_2}(x) := \left(1 - 3 \cdot \left(\frac{x}{\frac{l_1}{2}} \right)^2 + 2 \cdot \left(\frac{x}{\frac{l_1}{2}} \right)^3 \right) \cdot \text{dn_LinAlgEigenvectors}(A)_{1\ 2} + \left(\frac{x}{\frac{l_1}{2}} - 2 \cdot \left(\frac{x}{\frac{l_1}{2}} \right)^2 + \left(\frac{x}{\frac{l_1}{2}} \right)^3 \right) \cdot \frac{\frac{l_1}{2}}{m} \cdot \text{dn_LinAlgEigenvectors}(A)_{2\ 2}$$

$$w_{element2_2}(x) := w_{12_2} \left(x - \frac{l_1}{2} \right)$$

$$w_{21_2}(x) := \left(1 - 3 \cdot \left(\frac{x}{\frac{\alpha \cdot l_2}{m}} \right)^2 + 2 \cdot \left(\frac{x}{\frac{\alpha \cdot l_2}{m}} \right)^3 \right) \cdot \text{dn_LinAlgEigenvectors}(A)_{3\ 2} + \left(\frac{x}{\frac{\alpha \cdot l_2}{m}} - 2 \cdot \left(\frac{x}{\frac{\alpha \cdot l_2}{m}} \right)^2 + \left(\frac{x}{\frac{\alpha \cdot l_2}{m}} \right)^3 \right) \cdot \frac{\frac{\alpha \cdot l_2}{m}}{m} \cdot \text{dn_LinAlgEigenvectors}(A)_{4\ 2}$$

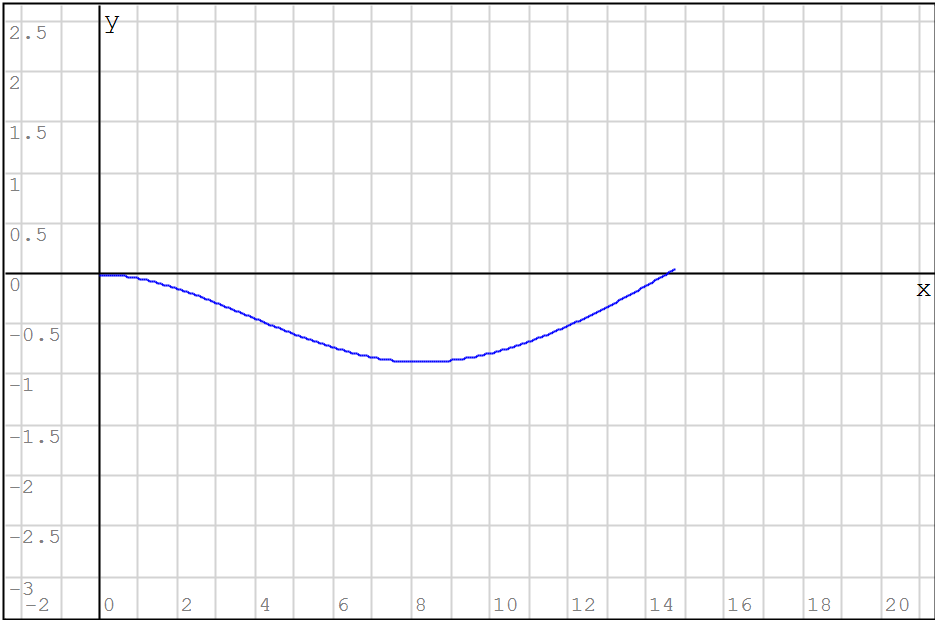
$$w_{element3_2}(x) := w_{21_2} \left(x - \frac{l_1}{m} \right)$$

$$w_{22_2}(x) := \left(1 - 3 \cdot \left(\frac{x}{\frac{(1-\alpha) \cdot l_2}{m}} \right)^2 + 2 \cdot \left(\frac{x}{\frac{(1-\alpha) \cdot l_2}{m}} \right)^3 \right) \cdot \text{dn_LinAlgEigenvectors}(A)_{5\ 2} + \left(\frac{x}{\frac{(1-\alpha) \cdot l_2}{m}} - 2 \cdot \left(\frac{x}{\frac{(1-\alpha) \cdot l_2}{m}} \right)^2 + \left(\frac{x}{\frac{(1-\alpha) \cdot l_2}{m}} \right)^3 \right) \cdot \frac{\frac{(1-\alpha) \cdot l_2}{m}}{m} \cdot \text{dn_LinAlgEigenvectors}(A)_{6\ 2}$$

$$(l_1 + \alpha \cdot l_2)$$

$$w_{element4_2}(x) := w_{22_2} \left(x - \frac{l_1}{m} \right)$$

$$w_{second}(x) := \text{if } (x \geq 0) \wedge \left(x < \frac{l_1}{m} \right) \\ w_{element1_2}(x) \\ \text{else} \\ \text{if } \left(x \geq \frac{l_1}{m} \right) \wedge \left(x \leq \frac{l_1}{m} \right) \\ w_{element2_2}(x) \\ \text{else} \\ \text{if } \left(x > \frac{l_1}{m} \right) \wedge \left(x \leq \frac{l_1 + \alpha \cdot l_2}{m} \right) \\ w_{element3_2}(x) \\ \text{else} \\ \text{if } \left(x > \frac{l_1 + \alpha \cdot l_2}{m} \right) \wedge \left(x \leq \frac{l_1 + l_2}{m} \right) \\ w_{element4_2}(x) \\ \text{else} \\ \text{"not defined"}$$



$w_{second}(x)$

$$w_{11_3}(x) := \left(3 \cdot \left(\frac{x}{\frac{l_1}{m}} \right)^2 - 2 \cdot \left(\frac{x}{\frac{l_1}{m}} \right)^3 \right) \cdot \text{dn_LinAlgEigenvectors}(A)_{13} + \left(- \left(\frac{x}{\frac{l_1}{m}} \right)^2 + \left(\frac{x}{\frac{l_1}{m}} \right)^3 \right) \cdot \frac{l_1}{m} \cdot \text{dn_LinAlg}$$

$$w_{element1_3}(x) := w_{11_3}(x)$$

$$w_{12_3}(x) := \left(1 - 3 \cdot \left(\frac{\frac{x}{l_1}}{\frac{2}{m}} \right)^2 + 2 \cdot \left(\frac{\frac{x}{l_1}}{\frac{2}{m}} \right)^3 \right) \cdot \text{dn_LinAlgEigenvectors}(A)_{1\ 3} + \left(\frac{\frac{x}{l_1}}{\frac{2}{m}} - 2 \cdot \left(\frac{\frac{x}{l_1}}{\frac{2}{m}} \right)^2 + \left(\frac{\frac{x}{l_1}}{\frac{2}{m}} \right)^3 \right) \cdot \frac{\frac{l_1}{2}}{m} \cdot$$

$$w_{element2_3}(x) := w_{12_3} \left(x - \frac{l_1}{m} \right)$$

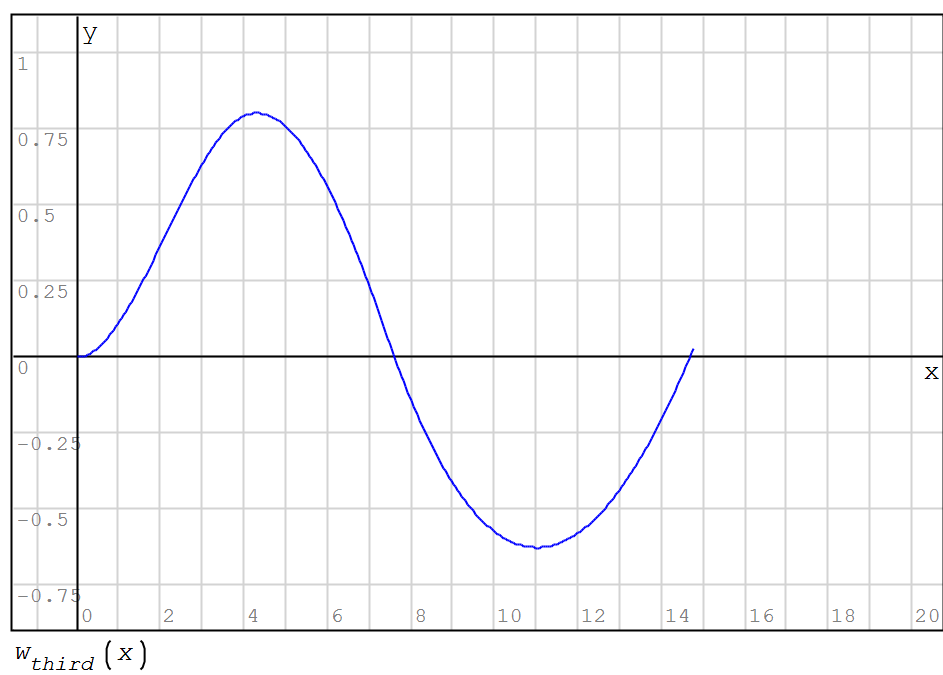
$$w_{21_3}(x) := \left(1 - 3 \cdot \left(\frac{\frac{x}{\alpha \cdot l_2}}{\frac{m}{m}} \right)^2 + 2 \cdot \left(\frac{\frac{x}{\alpha \cdot l_2}}{\frac{m}{m}} \right)^3 \right) \cdot \text{dn_LinAlgEigenvectors}(A)_{3\ 3} + \left(\frac{\frac{x}{\alpha \cdot l_2}}{\frac{m}{m}} - 2 \cdot \left(\frac{\frac{x}{\alpha \cdot l_2}}{\frac{m}{m}} \right)^2 + \left(\frac{\frac{x}{\alpha \cdot l_2}}{\frac{m}{m}} \right)^3 \right) \cdot \frac{\frac{l_1}{2}}{m} \cdot$$

$$w_{element3_3}(x) := w_{21_3} \left(x - \frac{l_1}{m} \right)$$

$$w_{22_3}(x) := \left(1 - 3 \cdot \left(\frac{\frac{x}{(1-\alpha) \cdot l_2}}{\frac{m}{m}} \right)^2 + 2 \cdot \left(\frac{\frac{x}{(1-\alpha) \cdot l_2}}{\frac{m}{m}} \right)^3 \right) \cdot \text{dn_LinAlgEigenvectors}(A)_{5\ 3} + \left(\frac{\frac{x}{(1-\alpha) \cdot l_2}}{\frac{m}{m}} - 2 \cdot \left(\frac{\frac{x}{(1-\alpha) \cdot l_2}}{\frac{m}{m}} \right)^2 + \left(\frac{\frac{x}{(1-\alpha) \cdot l_2}}{\frac{m}{m}} \right)^3 \right) \cdot \frac{\frac{l_1}{2}}{m} \cdot$$

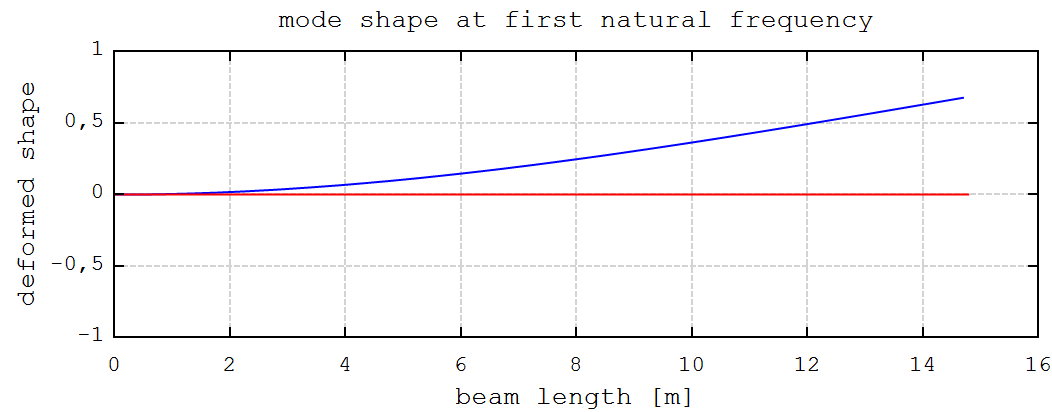
$$w_{element4_3}(x) := w_{22_3} \left(x - \frac{l_1 + \alpha \cdot l_2}{m} \right)$$

$$w_{third}(x) := \text{if } (x \geq 0) \wedge \left(x < \frac{l_1}{m} \right) \\ w_{element1_3}(x) \\ \text{else} \\ \text{if } \left(x \geq \frac{l_1}{m} \right) \wedge \left(x \leq \frac{l_1}{m} \right) \\ w_{element2_3}(x) \\ \text{else} \\ \text{if } \left(x > \frac{l_1}{m} \right) \wedge \left(x \leq \frac{l_1 + \alpha \cdot l_2}{m} \right) \\ w_{element3_3}(x) \\ \text{else} \\ \text{if } \left(x > \frac{l_1 + \alpha \cdot l_2}{m} \right) \wedge \left(x \leq \frac{l_1 + l_2}{m} \right) \\ w_{element4_3}(x) \\ \text{else} \\ \text{"not defined"}$$



$$UNDEFORMD := \begin{bmatrix} l_1 + l_2 & 0 \\ 0 & 0 \end{bmatrix}$$

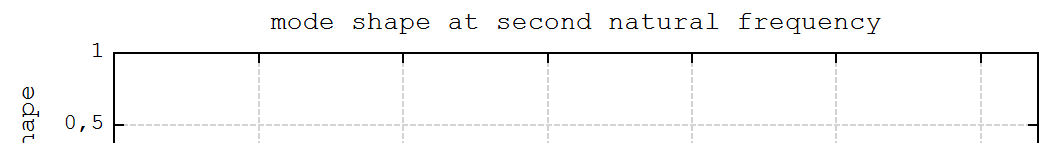
MODE SHAPES and NATURAL FREQUENCIES



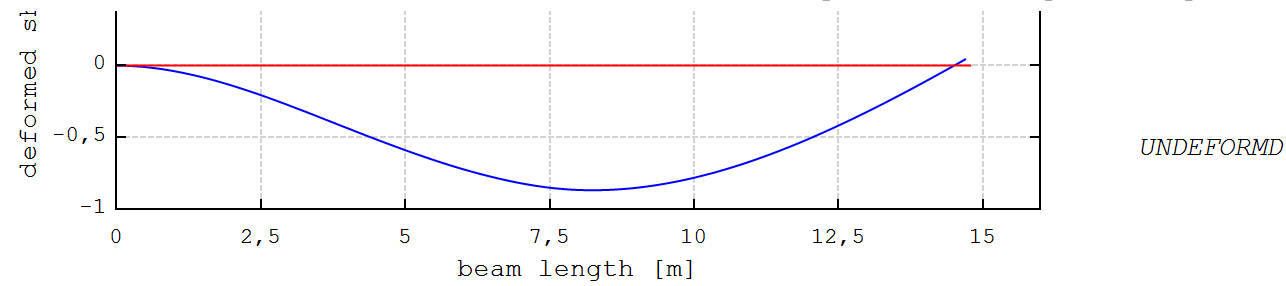
$$f_1 = 0.669 \text{ Hz}$$

$$UNDEFORMD = \begin{bmatrix} 14.8 \text{ m} & 0 \\ 0 & 0 \end{bmatrix}$$

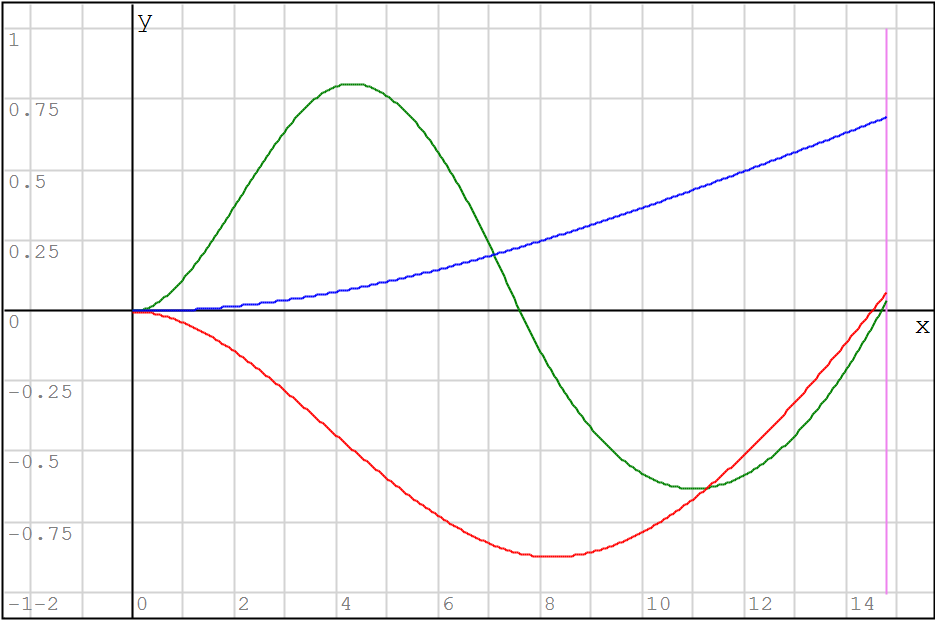
$$\begin{cases} w_{first}(x) \\ UNDEFORMD \end{cases}$$



$$f_2 = 11.137 \text{ Hz}$$



$$\begin{cases} w_{second}(x) \\ UNDEFORMD \end{cases}$$



$$\begin{cases} w_{first}(x) \\ w_{second}(x) \\ w_{third}(x) \\ \begin{bmatrix} 14.8 & 1 \\ 014.8 & -1 \end{bmatrix} \end{cases}$$